

# Document made available under the Patent Cooperation Treaty (PCT)

International application number: PCT/US04/043915

International filing date: 29 December 2004 (29.12.2004)

Document type: Certified copy of priority document

Document details: Country/Office: US  
Number: 60/533,586  
Filing date: 31 December 2003 (31.12.2003)

Date of receipt at the International Bureau: 31 January 2005 (31.01.2005)

Remark: Priority document submitted or transmitted to the International Bureau in compliance with Rule 17.1(a) or (b)



World Intellectual Property Organization (WIPO) - Geneva, Switzerland  
Organisation Mondiale de la Propriété Intellectuelle (OMPI) - Genève, Suisse



# THE UNITED STATES OF AMERICA

~~TO ALL TO WHOM THESE PRESENTS SHALL COME:~~

UNITED STATES DEPARTMENT OF COMMERCE

United States Patent and Trademark Office

*January 19, 2005*

THIS IS TO CERTIFY THAT ANNEXED HERETO IS A TRUE COPY FROM THE RECORDS OF THE UNITED STATES PATENT AND TRADEMARK OFFICE OF THOSE PAPERS OF THE BELOW IDENTIFIED PATENT APPLICATION THAT MET THE REQUIREMENTS TO BE GRANTED A FILING DATE.

APPLICATION NUMBER: 60/533,586

FILING DATE: *December 31, 2003*

RELATED PCT APPLICATION NUMBER: PCT/US04/43915



Certified by

*Don W. Dudas*

Under Secretary of Commerce  
for Intellectual Property  
and Director of the United States  
Patent and Trademark Office

Under the Paperwork Reduction Act of 1995, no persons are required to respond to a collection of information unless it displays a valid OMB control number.

**PROVISIONAL APPLICATION FOR PATENT COVER SHEET**

This is a request for filing a PROVISIONAL APPLICATION FOR PATENT under 37 CFR 1.53(c).

Express Mail Label No. EV 405538906 US

031451 U.S. PTO  
60/533586

INVENTOR(S)					
Given Name (first and middle (if any))		Family Name or Surname		Residence (City and either State or Foreign Country)	
Donald E.		Scott		Taupo, New Zealand	
Additional inventors are being named on the _____ separately numbered sheets attached hereto					
TITLE OF THE INVENTION (500 characters max)					
DIMENSIONALLY STABLE FABRIC					
Direct all correspondence to: CORRESPONDENCE ADDRESS					
<input type="checkbox"/> Customer Number: <span style="border: 1px solid black; display: inline-block; width: 200px; height: 30px; vertical-align: middle;"></span>					
OR					
<input checked="" type="checkbox"/> Firm or Individual Name		Chase Law Firm, L.C.			
Address		4400 College Blvd., Suite 130			
Address					
City	Overland Park	State	KS	Zip	66211
Country	United States of America	Telephone	913-339-9686	Fax	913-339-6061
ENCLOSED APPLICATION PARTS (check all that apply)					
<input checked="" type="checkbox"/> Specification Number of Pages 9		<input type="checkbox"/> CD(s), Number _____			
<input type="checkbox"/> Drawing(s) Number of Sheets _____		<input type="checkbox"/> Other (specify) _____			
<input type="checkbox"/> Application Date Sheet. See 37 CFR 1.76					
METHOD OF PAYMENT OF FILING FEES FOR THIS PROVISIONAL APPLICATION FOR PATENT					
<input checked="" type="checkbox"/> Applicant claims small entity status. See 37 CFR 1.27.		FILING FEE Amount (\$) <div style="border: 1px solid black; width: 100px; height: 50px; margin: 10px auto; text-align: center;">80.00</div>			
<input checked="" type="checkbox"/> A check or money order is enclosed to cover the filing fees.					
<input checked="" type="checkbox"/> The Director is hereby authorized to charge filing fees or credit any overpayment to Deposit Account Number: 03-1425					
<input type="checkbox"/> Payment by credit card. Form PTO-2038 is attached.					
The invention was made by an agency of the United States Government or under a contract with an agency of the United States Government.					
<input checked="" type="checkbox"/> No.					
<input type="checkbox"/> Yes, the name of the U.S. Government agency and the Government contract number are: _____					

[Page 1 of 2]

Respectfully submitted,

SIGNATURE

TYPED or PRINTED NAME Sean T. Bradley

TELEPHONE 913-339-9666

Date December 31, 2003

REGISTRATION NO. 46,572

(if appropriate)

Docket Number: 3350

**USE ONLY FOR FILING A PROVISIONAL APPLICATION FOR PATENT**

This collection of information is required by 37 CFR 1.51. The information is required to obtain or retain a benefit by the public which is to file (and by the USPTO to process) an application. Confidentiality is governed by 35 U.S.C. 122 and 37 CFR 1.14. This collection is estimated to take 8 hours to complete, including gathering, preparing, and submitting the completed application form to the USPTO. Time will vary depending upon the individual case. Any comments on the amount of time you require to complete this form and/or suggestions for reducing this burden, should be sent to the Chief Information Officer, U.S. Patent and Trademark Office, U.S. Department of Commerce, P.O. Box 1450, Alexandria, VA 22313-1450. DO NOT SEND FEES OR COMPLETED FORMS TO THIS ADDRESS. SEND TO: Mail Stop Provisional Application, Commissioner for Patents, P.O. Box 1450, Alexandria, VA 22313-1450.

If you need assistance in completing the form, call 1-800-PTO-9199 and select option 2.

# FEE TRANSMITTAL for FY 2002

Patent fees are subject to annual revision.

TOTAL AMOUNT OF PAYMENT (\$ 80.00)

## Complete if Known

Application Number	
Filing Date	
First Named Inventor	DONALD E. SCOTT
Examiner Name	
Group Art Unit	
Attorney Docket No.	3350

## METHOD OF PAYMENT

1. ☒ The Commissioner is hereby authorized to charge indicated fees and credit any overpayments to:
- Deposit Account Number 03-1425
- Deposit Account Name Chase & Yakimo
- ☒ Charge any Additional Fee Required Under 37 CFR 1.16 and 1.17
- ☒ Applicant claims small entity status. See 37 CFR 1.27
2. ☒ Payment Enclosed:
- ☒ Check ☐ Credit card ☐ Money Order ☐ Other

## FEE CALCULATION

### 1. BASIC FILING FEE

Large Entity	Small Entity	Fee Code	Fee Code	Fee Description	Fee Paid
		(S)	(S)		
101	740	201	370	Utility filing fee	
108	330	206	165	Design filing fee	
107	510	207	255	Plant filing fee	
108	740	208	370	Rescue filing fee	
114	160	214	80	Provisional filing fee	80.00

SUBTOTAL (1) (\$ 80.00)

### 2. EXTRA CLAIM FEES

Total Claims	Extra Claims	Fee from below	Fee Paid
Independent Claims	-20** =	X	
Multiple Dependent Claims	-3** =	X	

Large Entity	Small Entity	Fee Code	Fee Code	Fee Description
		(S)	(S)	
103	18	203	9	Claims in excess of 20
102	84	202	42	Independent claims in excess of 3
104	280	204	140	Multiple dependent claim, if not paid
109	84	209	42	** Reissue independent claims over original patent
110	18	210	9	** Reissue claims in excess of 20 and over original patent

SUBTOTAL (2) (\$ )

\*or number previously paid, if greater; For Reissues, see above

## FEE CALCULATION (continued)

### 3. ADDITIONAL FEES

Large Entity	Small Entity	Fee Code	Fee Code	Fee Description	Fee Paid
		(S)	(S)		
105	130	205	65	Surcharge - late filing fee or oath	
127	50	227	25	Surcharge - late provisional filing fee or cover sheet	
139	130	139	130	Non-English specification	
147	2,520	147	2,520	For filing a request for ex parte reexamination	
112	920*	112	920*	Requesting publication of S/R prior to Examiner action	
113	1,840*	113	1,840*	Requesting publication of S/R after Examiner action	
115	110	215	55	Extension for reply within first month	
116	400	216	200	Extension for reply within second month	
117	920	217	460	Extension for reply within third month	
118	1,440	218	720	Extension for reply within fourth month	
128	1,960	228	980	Extension for reply within fifth month	
119	320	219	160	Notice of Appeal	
120	320	220	160	Filing a brief in support of an appeal	
121	280	221	140	Request for oral hearing	
138	1,510	138	1,510	Petition to institute a public use proceeding	
140	110	240	55	Petition to revive - unavoidable	
141	1,280	241	640	Petition to revive - unintentional	
142	1,280	242	640	Utility issue fee (or reissue)	
143	460	243	230	Design issue fee	
144	620	244	310	Plant issue fee	
122	130	122	130	Petitions to the Commissioner	
123	50	123	50	Processing fee under 37 CFR 1.17(f)	
128	180	128	180	Submission of Information Disclosure Stmt	
581	40	581	40	Recording each patent assignment per property (times number of properties)	
146	740	246	370	Filing a submission after final rejection (37 CFR § 1.129(a))	
149	740	249	370	For each additional invention to be examined (37 CFR § 1.129(b))	
179	740	279	370	Request for Continued Examination (RCE)	
169	900	169	900	Request for expedited examination of a design application	

Other fee (specify)

\*Reduced by Basic Filing Fee Paid

SUBTOTAL (3) (\$ ) -0-

## SUBMITTED BY

Name (Print/Type)	Sean T. Bradley	Registration No. (Attorney/Agent)	46,572	Telephone	913-339-9666
Signature		Date	12/31/2003		

WARNING: Information on this form may become public. Credit card information should not be included on this form. Provide credit card information and authorization on PTO-2038.

Burden Hour Statement: This form is estimated to take 0.2 hours to complete. Time will vary depending upon the needs of the individual case. Any comments on the amount of time you are required to complete this form should be sent to the Chief Information Officer, U.S. Patent and Trademark Office, Washington, DC 20231. DO NOT SEND FEES OR COMPLETED FORMS TO THIS ADDRESS. SEND TO: Assistant Commissioner for Patents, Washington, DC 20231.

## **DIMENSIONALLY STABLE FABRIC**

### **FIELD OF THE INVENTION**

This invention relates to wool-synthetic blend fabrics and more particularly to flame-resistant, dimensionally stable wool-synthetic blend fabrics suitable for use in  
5 aircraft interiors.

### **BACKGROUND OF THE INVENTION**

Upholstery fabrics made from wool are known to have an attractive appearance and feel to the touch. Due to the tendency of wool to shrink after washing in water, however, attempts have been made to substitute wool fabrics with fabrics made from  
10 synthetic materials such as polyester. The appearance and feel of fabrics made from synthetic materials, however, has been found to be inferior to that of fabrics made from wool. Fabrics made from blends of wool fibers with certain synthetic fibers retain some of the aesthetic features of wool as well as some of the cost benefits and potential property advantages of synthetics.

15 In the aircraft industry, seat cover fabrics are subject to specifications provided by aircraft manufacturers such as Airbus and Boeing. The relevant Airbus technical specification, for example, is TL 25/5092/83. The relevant flammability, smoke and toxicity portions of the standard are FAR 25.853 (b), appendix F, amended 32, JAR 25853 (b), appendix change 10, and ABD 0031 (previously numbered ATS 1000.001). These  
20 specifications include standards for abrasion resistance including resistance to abrasion simulated by a Martindale tester. Resistance to stains resulting from spills, and to loss of color and shrinkage due to washing, is also specified. Seat cover fabrics may be required to meet specifications after a minimum of 10 washings. An areal weight below 470 g/m<sup>2</sup> is (Docket 3350)

specified. It is desirable that shrinkage during service life, including shrinkage due to cleaning processes, be minimized. Resistance to pilling, corrosion and color loss may also be specified.

The relevant Boeing specification is BMS 8-236, for general upholstery interior applications. The flammability standard is provided by BSS7230, a twelve second vertical burn test, in which the sample is required to self extinguish within fifteen seconds, with a burn length of less than eight inches. Drips, if any, are required to extinguish in less than five seconds. Smoke emissions of less than 200 are specified according to BSS7238. Prescribed limits for individual toxic components in toxic gas emissions are tested according to BSS 7239. Dimensional stability is evaluated after prescribed cleaning, whether dry cleaning or water washing methods are used. While zero shrinkage is ideal, shrinkage levels of less than 6%, in both warp and fill directions, are acceptable. Standards for appearance, snag resistance, pilling resistance, color fastness and strength are part of the overall specification.

Wool fabrics are typically cleaned using a dry-cleaning process, including immersion in a solvent such as perchloroethylene, in order to maintain the dimensional stability of the fabric. Due to environmental and cost considerations, it would be desirable to clean wool-based fabrics without the use of perchloroethylene or other organic solvents. Water containing surfactants or detergents is highly effective in cleaning such fabrics, however, use of water-based cleaning solutions has been limited by the tendency of wool based fabrics to shrink after being subjected to such solutions. Synthetic fibers, on the other hand, are typically highly resistant to shrinkage following washing in water. Synthetic fibers, however, tend to be highly flammable.

Because of the nature of the constituent parts of the above mentioned wool-synthetic blends, such blends in the prior art are typically neither flame resistant, nor shrink resistant when washed in water. There is a need for fabrics made from wool-synthetic blends that will meet the special requirements for aircraft interiors.

5

### BRIEF DESCRIPTION OF THE INVENTION

In one embodiment of the invention, a method of producing a dimensionally stable, fire-resistant fabric suitable for use on aircraft includes the steps of providing a yarn having a blend of wool fibers and fire-resistant synthetic fibers, the wool  
10 fibers comprising approximately 30% to 70% of the blend, weaving the yarn to form a fabric, and heat setting the fabric to achieve a washable woven structure resistant to shrinkage. The synthetic fibers may include polyester fibers produced or treated to enhance fire resistance.

In another embodiment a method is provided for producing a dimensionally stable, fire-resistant fabric by spinning wool and fire-resistant polyester fibers to form a yarn,  
15 weaving the yarn to form a fabric, and heat-setting the fabric to produce a finished material that passes Airbus and/or Boeing specifications.

In a further embodiment a method is provided for producing a fire-resistant wool-based yarn by spinning shortened wool fibers with fire-resistant polyester fibers in a vortex spinning apparatus, weaving the yarn into an aircraft manufacturer-approved fabric,  
20 and heat-setting the fabric in a stenter apparatus such that it resists shrinkage after water washing. In a further embodiment, the method includes treating the yarn or fabric with zirconium to augment the fire-resistant properties.

In yet another embodiment a method is provided for producing a dimensionally stable fabric by providing wool fibers, an effective percentage thereof cut or broken to fall within a selected length range, providing fire-resistant synthetic fibers, spinning the wool and synthetic fibers to produce a wool-synthetic blend yarn, wherein the wool fibers  
5 comprise approximately 30% to 70% of the blend, weaving the yarn to form a fabric, and heat setting the fabric to produce a fabric that passes aircraft manufacturer specifications.

Wool fibers having a typical length of no greater than approximately five centimeters may be prepared by stretch-breaking. The synthetic fibers may include polyester fibers. Fibers may be spun by delivering the fibers to a ring spinning, air-jet spinning or  
10 vortex spinning apparatus for spinning the fibers into a yarn. The fabric may be heat-set by securing and heating the fabric within a stenter. When passing the fabric through a stenter, sufficient heat is applied to set the fabric and produce a dimensionally stabilized fabric resistant to shrinkage. Further steps may include applying zirconium fire retardant to the fabric and applying a coating to bind the zirconium fire retardant to the fabric.



## DETAILED DESCRIPTION

In one embodiment, wool fibers are first prepared by reducing their length.

Wool tops, consisting of fibers that are approximately 5.5 to 8 cm in length, are passed through a stretch-breaking machine to reduce their lengths to approximately 2 to 5 cm. It is advantageous if the fibers are approximately 3 to 4 cm in length. It is advantageous if the wool fibers have diameters in the range of 17 to 25 microns, and particularly advantageous if the wool fibers have diameters in the range of approximately 22 to 25 microns.

After stretch breaking, the wool fibers are combined with flame retardant (FR) polyester fibers having a length of approximately 2 to 5 cm, and the resulting combined fiber bundles are passed through one or more draw frames. The drafted wool and FR polyester fiber bundles are introduced into a spinning machine at such relative rates as to achieve wool contents in the range of approximately 30 to 70 percent. It is advantageous to the properties of the resulting fabric if the wool content is in the range of approximately 40 to 60 percent.

### 15 Spinning Technology

Typically, carding occurs prior to, or as an initial step in, the spinning process. Through carding, fibers are straightened and made relatively parallel to one another. After carding the fibers form a thin layer called a web. The web is gathered into a loose rope called a sliver. The sliver is typically wound into a large can and then moved to a draw frame. In the drawing process, multiple cans of sliver are drawn together to form a combined sliver.

Ring spinning is a relatively slow spinning technology that typically yields a high quality yarn. During ring spinning, sliver is fed into the drafting zone of the ring spinning frame. The drafting zone has one roller that turns relatively slow and feeds the

sliver and another roller that turns relatively fast. The faster roller pulls out a few fibers at a time forming a fine stream of fibers that are fed to a rotating spindle inside a ring. As the spindle rotates, it drags a slower moving traveler on the ring. The ring twists the fibers as they are wound onto a bobbin that rides on the spindle. After spinning, the yarn may then be  
5 used for weaving, perhaps after being further transferred to other holding structures. Ring spinning has been the preferred method of producing high quality wool yarns that demonstrate superior feel to the touch and abrasion resistance.

The air-jet spinning method uses air currents to twist fibers together, resulting in higher throughput and productivity than ring spinning. Air-jet spinning may be  
10 used to spin blends of wool and FR polyester fibers, but yields yarns with reduced abrasion resistance in comparison with ring and vortex spinning.

The air vortex spinning method is a particularly efficient spinning method that is capable of spinning yarns at very high speeds and that yields a yarn having a relatively smooth texture and increased abrasion resistance. A vortex spinning apparatus  
15 typically takes drawn sliver and drafts it to the desired yarn count via a four-roller drafting unit. The drafted fibers are then sucked into a nozzle where a high speed air vortex wraps the fibers around the outside of a hollow stationary spindle. Yarn twist is then imparted as the fibers are pulled down a shaft that runs through the middle of the spindle.

An example of a vortex spinning apparatus is described in the patent to  
20 Mori, patent No. 6,370,858, hereby incorporated by reference. Mori discloses a vortex spinning method in which a drafted fiber bundle is supplied to a nozzle block and then to a hollow guide shaft. A core fiber is also fed to the nozzle block and then to the hollow guide shaft. Vortex air currents ejected from spinning nozzles in the nozzle block cause inversely turned fibers to wrap the fiber bundle and core fiber to create core yarn. The core fiber may

be multi-filament in which case the vortex air currents balloon the multiple filaments, resulting in the filaments being partially separated from one another. The vortex air currents insert the front ends of the fibers into the clearances between the separated filaments, and cause the other ends of the fibers to wrap around the multi-filament core fiber, resulting in the  
5 creation of the core yarn.

The spinning speed of a vortex spinner is much faster than that provided by ring spinning with the ring method typically producing yarn at the rate of 20 meters per minute and the vortex method typically producing yarn at the rate of 400 meters per minute. The vortex method does not readily accommodate the longer fibers typically used in wool  
10 spinning, however, and it has been found to be advantageous to reduce the fiber lengths as illustrated in the various embodiments of the invention disclosed herein.

#### Preparation of the Fabric

In the various embodiments contained herein, the spinning process used to produce the yarn may include ring spinning, air-jet spinning, air vortex spinning or other  
15 appropriate means. It is advantageous, however, to spin the yarn using a vortex spinning method and apparatus.

After spinning, the yarn is typically dyed to a selected color and then woven into a fabric. The particular weave is typically determined by the requirements of the eventual use of the fabric. Appropriate weaves include those known for use by American  
20 Airlines and United Airlines.

After weaving, the fabric is heat-set to increase dimensional stability of the fabric. It is advantageous if the heat setting includes the step of affixing the fabric within a stenter frame so that a given dimension may be controlled during the heat-setting process. The fabric is heat set within the stenter by heating the fabric to a temperature in excess of

100°C. The actual temperature used is primarily dependent upon the chemical nature of the synthetic fiber being used. Multiple heating bays may be used, each successive bay typically providing increased heat. In the case where a polyester fiber is used, the maximum temperature is typically set between approximately 170°C and 220°C. Dwell time, the time  
5 period in which heat is applied to the fabric in the stenter may be adjusted according to temperatures used and composition of the fabric. The fabric is typically heated by provision of dry heat using appropriate means such as a gas fired burner and heat exchanger. In one embodiment, dimensional stability results from incipient melting of polyester (or other synthetic) fibers and subsequent bonding of the fibers to form a continuous or semi-  
10 continuous polyester network or lattice within the fabric.

In an embodiment directed to vortex spun yarn, wool tops are passed through a stretch-breaking apparatus and the fiber length is thereby reduced to approximately 3 to 4 cm. The wool fibers are then combined with FR polyester staple having an approximate length of 3 cm, at a ratio of one part wool fiber to one part FR polyester. The  
15 combined fibers are drafted on a drawframe and then spun in a vortex spinner. Portions of the yarn are dyed to a desired color or colors and then woven into a fabric suitable for use in aircraft such as for seat upholstery. The fabric is heat set in a stenter at approximately 190°C for approximately 30 seconds. As a result of this process the fabric meets airline interior fabric test specifications, including those for fire resistance, abrasion and shrinkage after  
20 water washing. By way of example, a fabric may be produced in accordance with the above embodiment to pass Airbus specification TL 25/5092/83 and Boeing specification BMS 8-236.

Additionally, yarn spun from a blend of wool and synthetic fibers, the wool fibers comprising between approximately 30 to 70 percent of the blend, may be treated with

zirconium-based fire retardants prior to weaving to augment the fire-resistant qualities of the resulting fabric. The zirconium treatment may also be applied to any of the other fabrics set forth above.

- To resist dislodging of the zirconium fire retardant from the fabric during
- 5 washing, the fabric may be treated with polyurethane or other appropriate material to coat the zirconium and bind it to the fabric.